

# **Study of Swirling Jet on an Impingement Surface**

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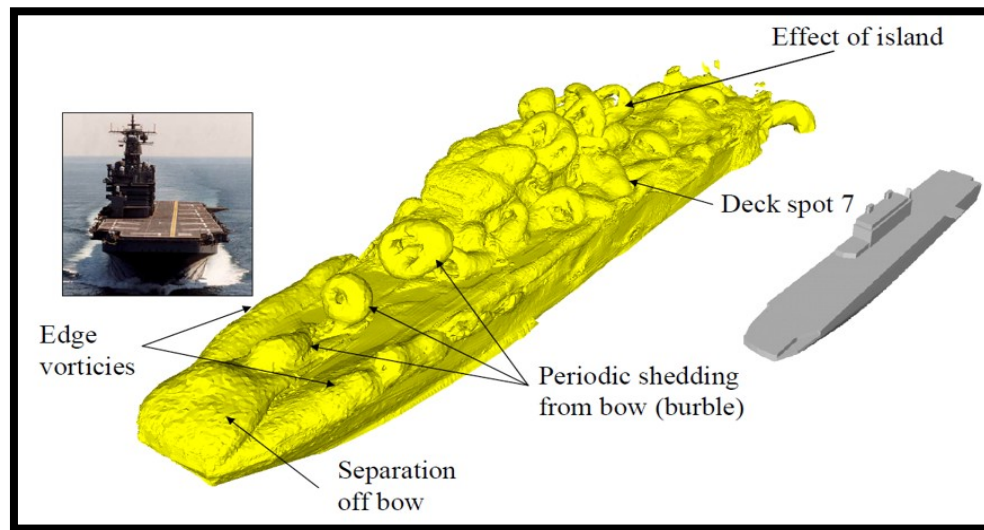
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# Background

- Vortex flow is seen in many applications:
  - “Burple” effect on aircraft carriers [1]
  - Vortex wake at airports
- Impingement wall could be a potential solution to reducing vortex wake and “burble” effect
- How does vortex flow interact with angles impingement surfaces?



Airflow Simulation of Aircraft Carrier [13]

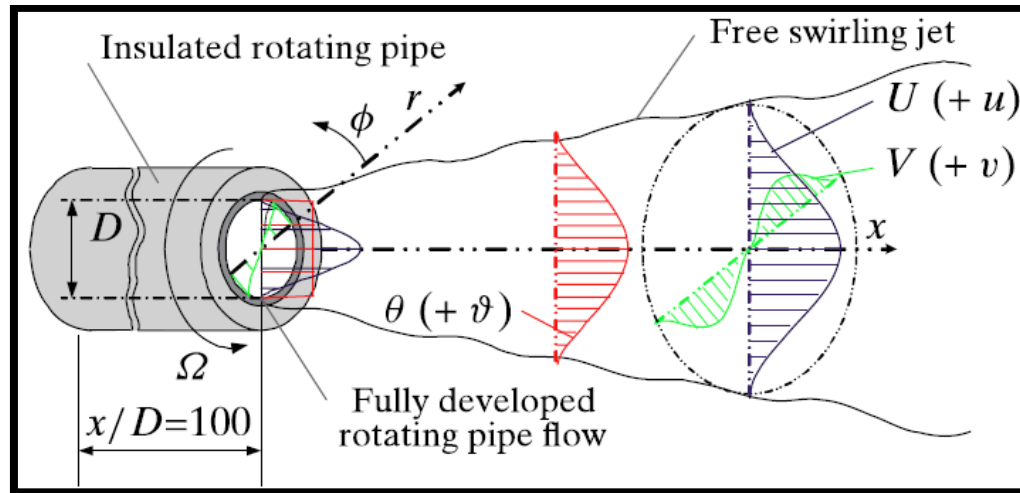
# Physics of Jets and Vortex Flow

Incompressible Navier – Stokes equation:  $\frac{1}{r} \frac{\partial}{\partial r} (rW) + \frac{\partial U}{\partial x} = 0$

Swirl Intensity:  $S_{\phi x} = \left[ \int_0^{\infty} r^2 UV dr \right] \left[ R \int_0^{\infty} r \left( U^2 - \frac{1}{2} V^2 \right) dr \right]^{-1}$

Reynolds Number:  $Re = \frac{U * D}{\nu}$

Swirl Number:  $S = \frac{V}{U}$

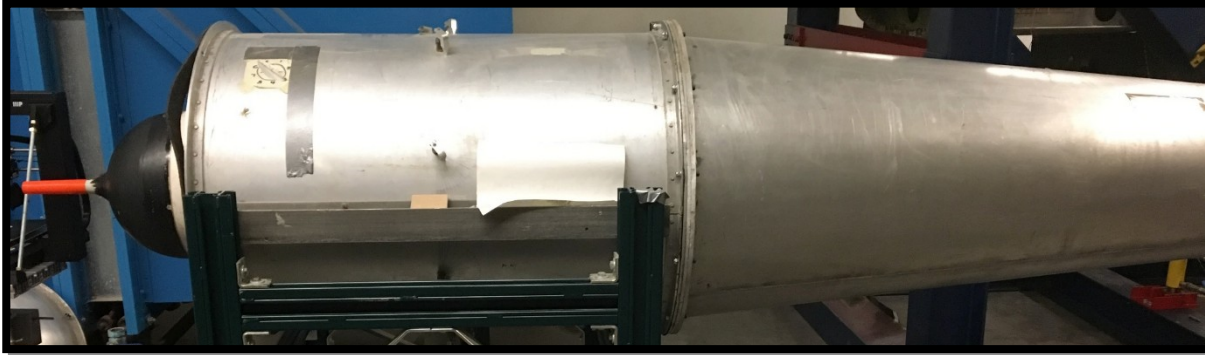


Free Flow Developing Swirling Jet [12]

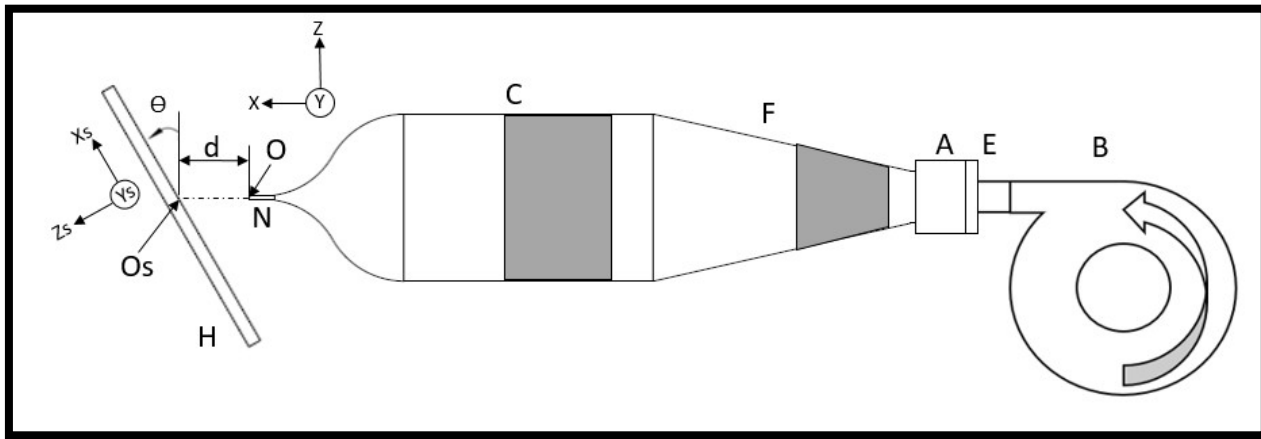
# Research Objectives

- Compare diverging/converging blower system to previous studies to validate setup and instrumentation methods.
  - Measure axial jet stream profile of tripped nozzle setup
- Develop flow visualization technique to capture flow field interactions
- Develop three swirling jet stream nozzles with varying swirl number
  - Characterize axial and azimuthal profiles of the unimpeded flow for each nozzle
- Characterize flow interaction for tripped nozzle and three swirling jet stream nozzles on the impingement wall at three different angles and Reynolds numbers

# Experimental Facilities



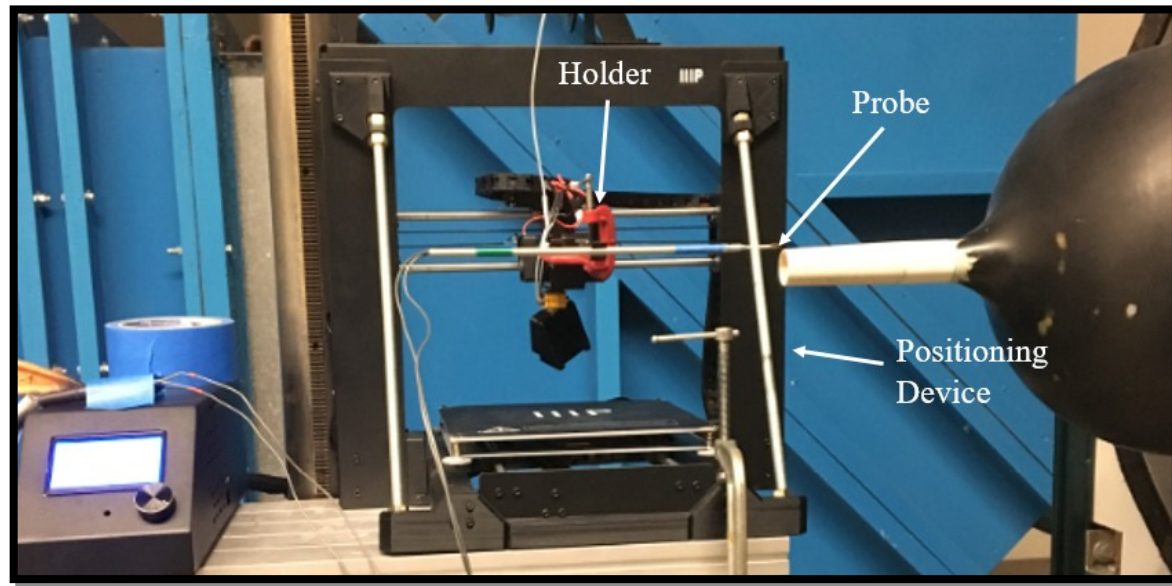
Facility jet system; Micro Air Vehicles Laboratory,  
AME Building, University of Arizona



Schematic of jet system

# Hot Wire Measurement Technique

- Axial and azimuthal measurements taken for 1 second at 1kHz sample rate along the centerline of the nozzle at discrete location 2mm apart.
- Each data set was averaged to give a single data point for the discrete location



Hot Wire Measurement Setup

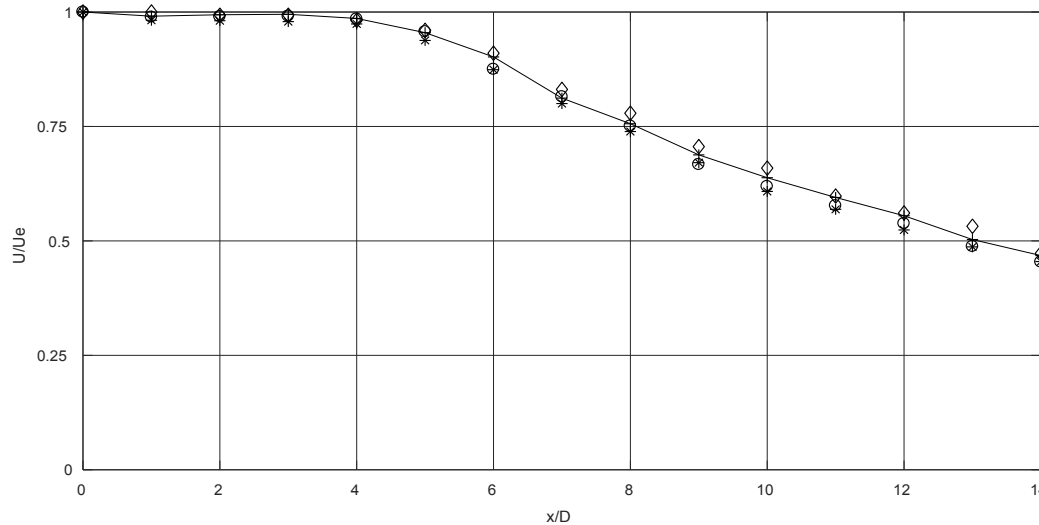
# Tripped Nozzle Experiment



Boundary lay trip ring 3D printed and installed in nozzle at 1" from exit

- PLA Model 100% infill
- 0.140" length and .020" thick with four axial notches [6]
- Axial jet stream profile measured at 5 stations spaced two inch apart; Centerline profile measured to  $15D$
- Measured for four Reynolds numbers ,  $6.2 \times 10^4$ ,  $8.30 \times 10^4$ ,  $1.02 \times 10^5$ ,  $1.24 \times 10^5$

# Tripped Nozzle Mean Axial Speed on the Centerline

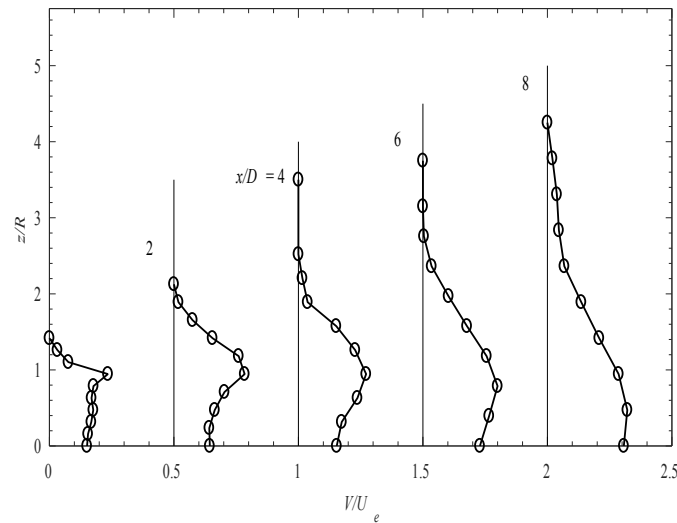
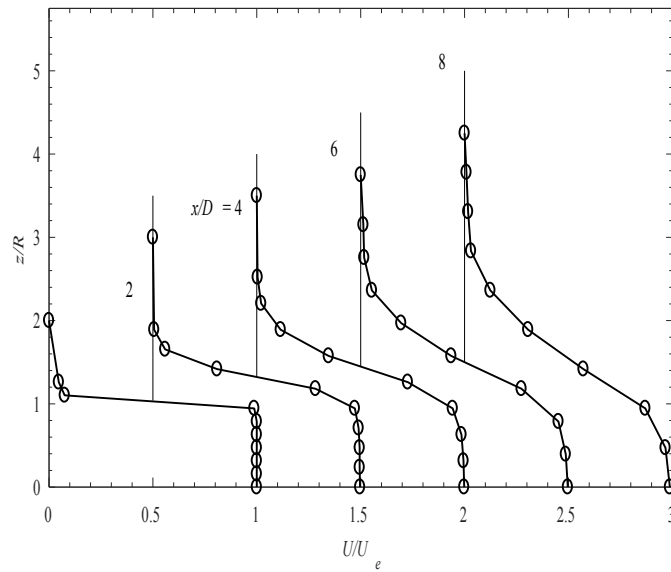


Profile of the mean axial speed on the centerline at several Reynolds numbers, as denoted by the following data symbols:  $\diamond$ ,  $6.2 \times 10^4$ ;  $+$ ,  $8.30 \times 10^4$ ;  $\circ$ ,  $1.02 \times 10^5$ ;  $*$ ,  $1.24 \times 10^5$ , --, [6].

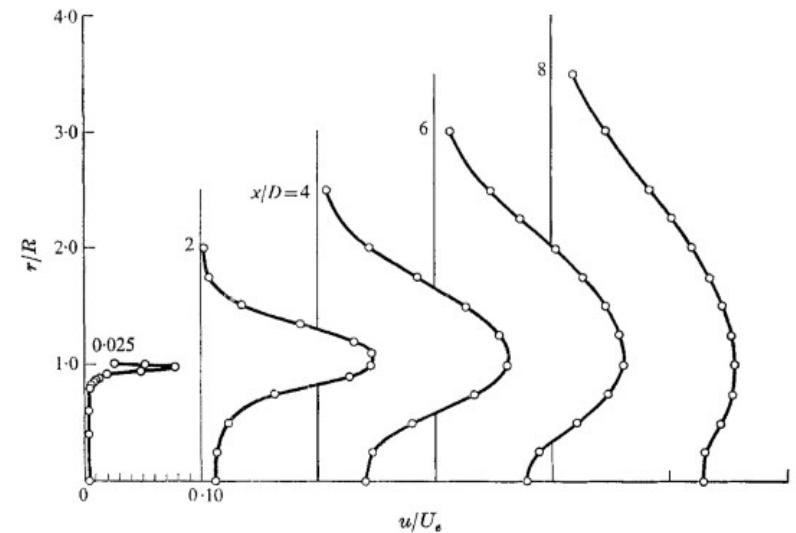
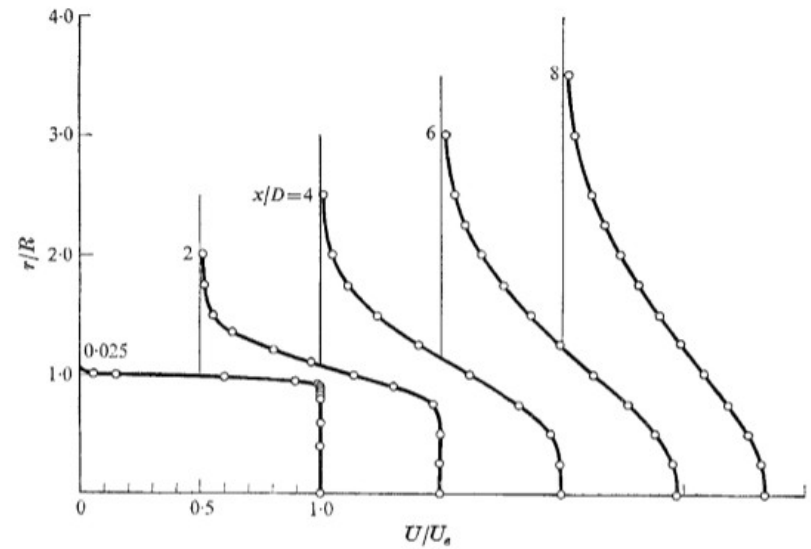
- Centerline profile does not change with Re
- Compares closely to pervious research [6]



# Axial and Azimuthal Profile

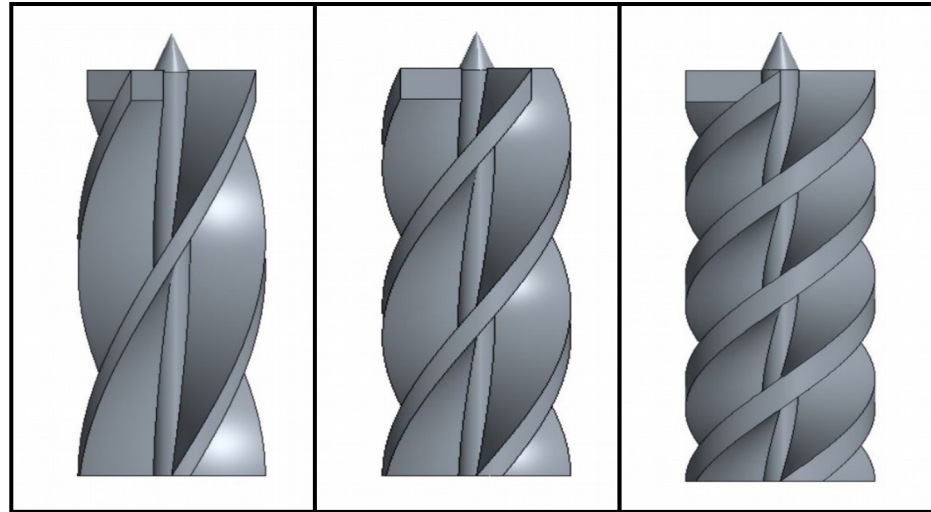


Measured profiles from blower



Profiles from previous research [6]

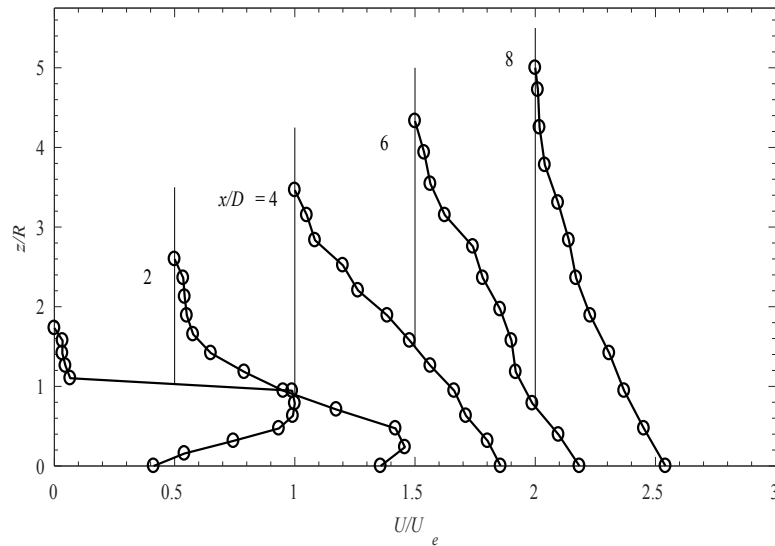
# Swirling Nozzle Characterization



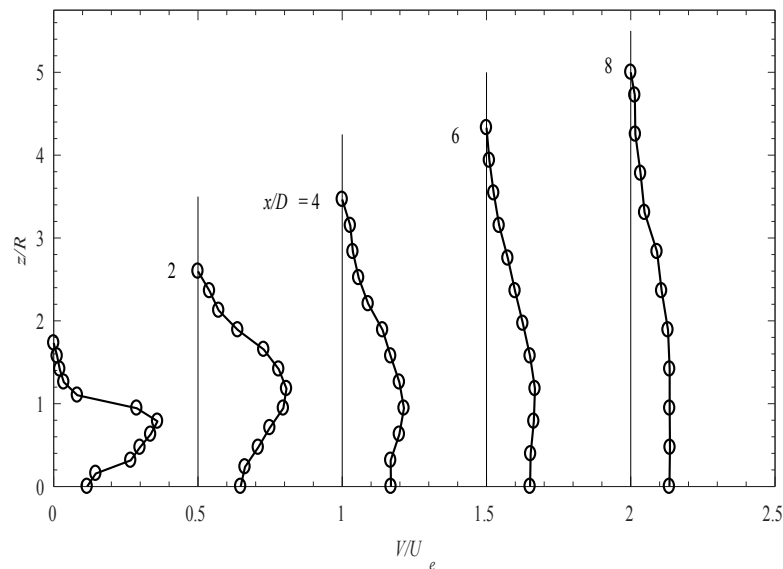
Interior of 3D printed Swirling Jet Steam Nozzles

- PLA Model 100% infill
- Nozzle's blades were swept at 3.00", 5.00" and 7.00" pitch
- Axial jet stream profile measured at 5 stations spaced two inch apart;
- Measured at Reynolds number  $6.2 \times 10^4$

# Nozzle 3 Axial and Azimuthal Profile



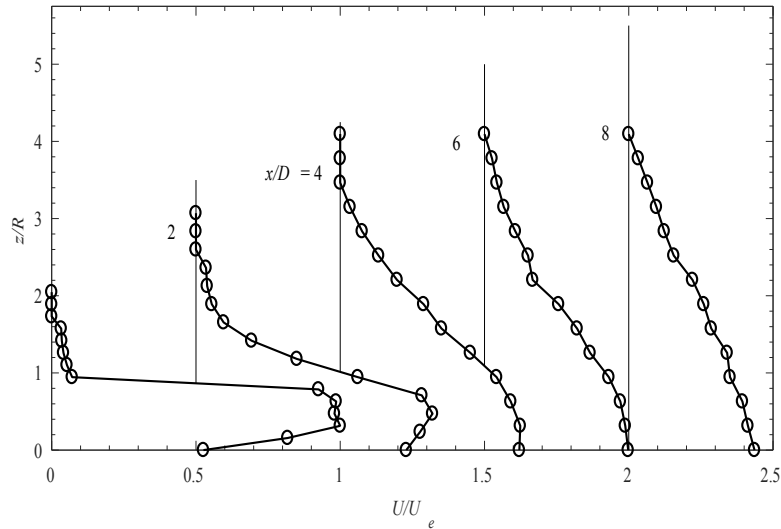
Axial Profile



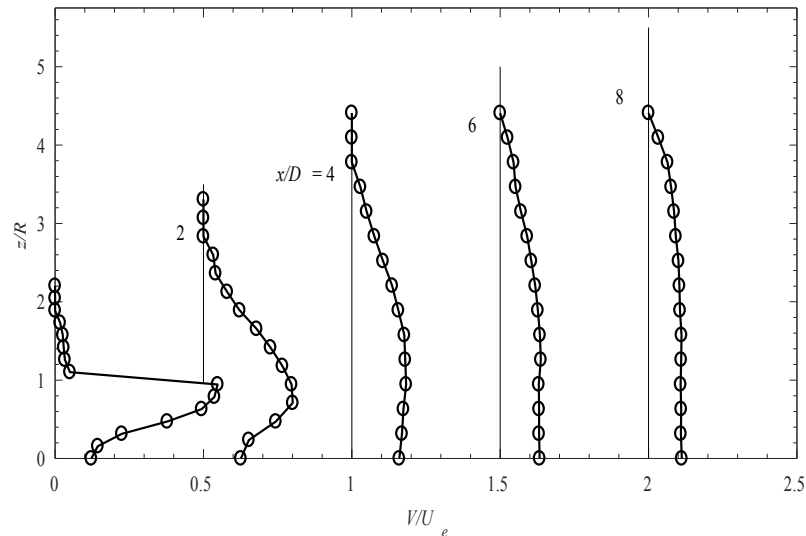
Azimuthal Profile

- Similar profile to tripped nozzle jet stream
- Axial become a conical shape
- Azimuthal profile dissipates quicker

# Nozzle 5 Axial and Azimuthal Profile



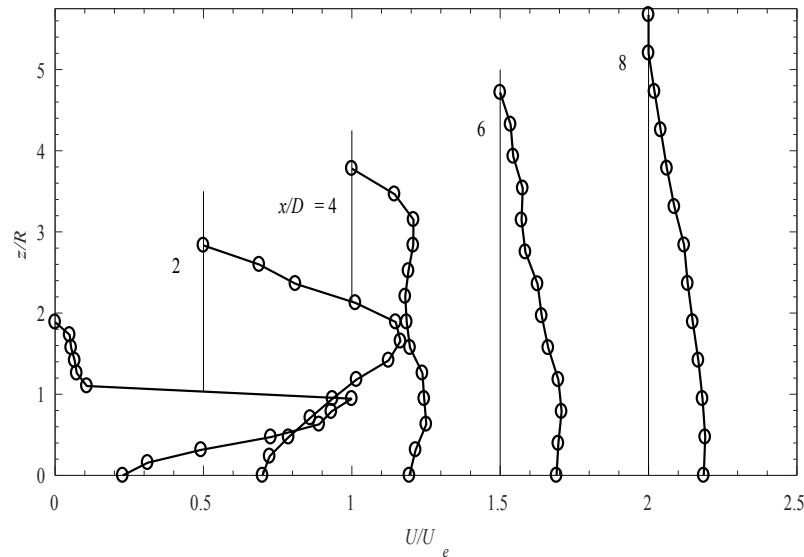
Axial Profile



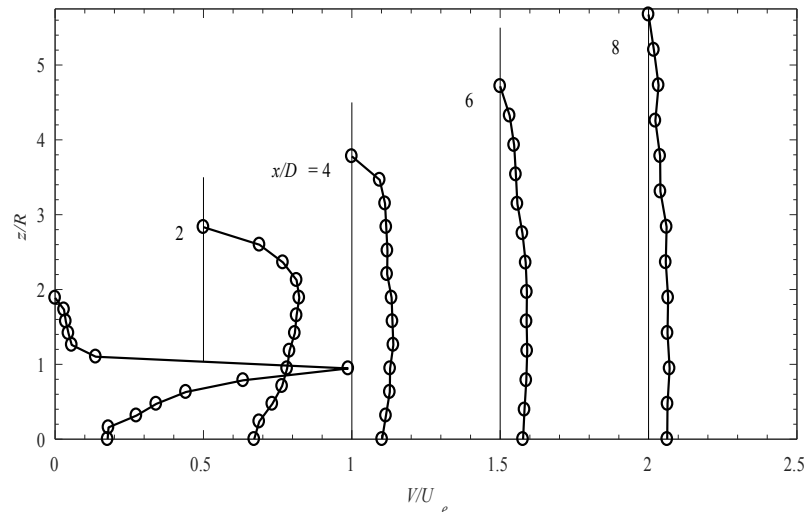
Azimuthal Profile

- Similar profile to nozzle 3 jet stream
- There is less of an axial component than nozzle 3
- Azimuthal profile is stronger at the first two stations

# Nozzle 7 Axial and Azimuthal Profile



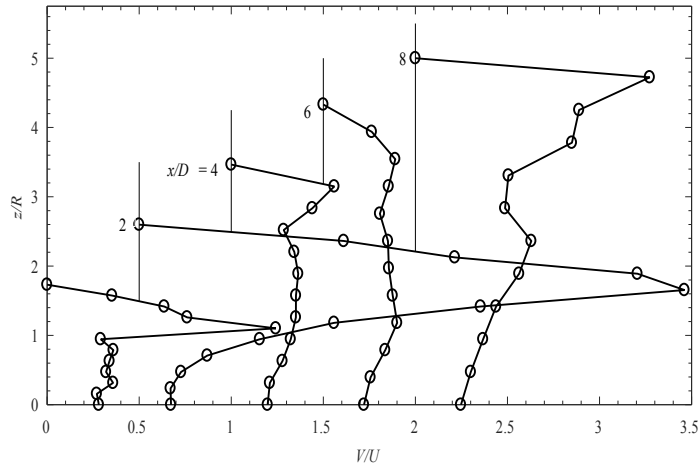
Axial Profile



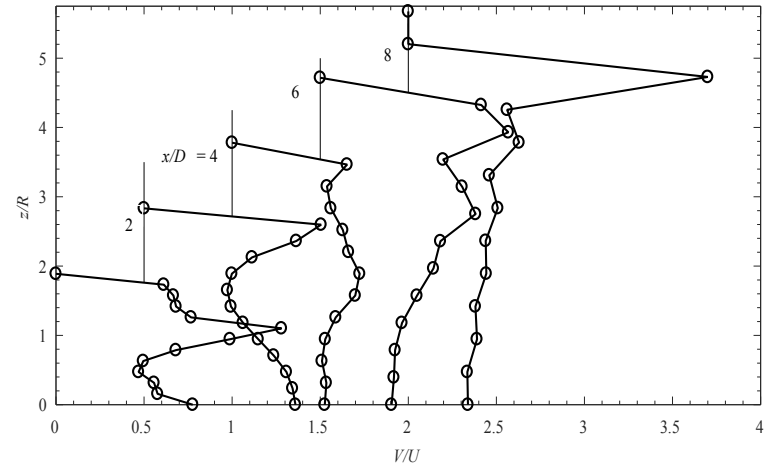
Azimuthal Profile

- Irregular axial profile
- Axial profile resembles the azimuthal profile for nozzle 5
- Azimuthal profile is strongest at the first two stations

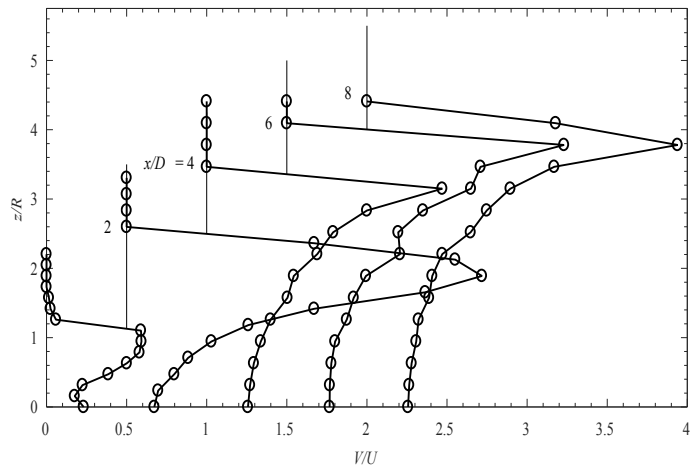
# Swirling Intensity Profiles



Nozzle 3



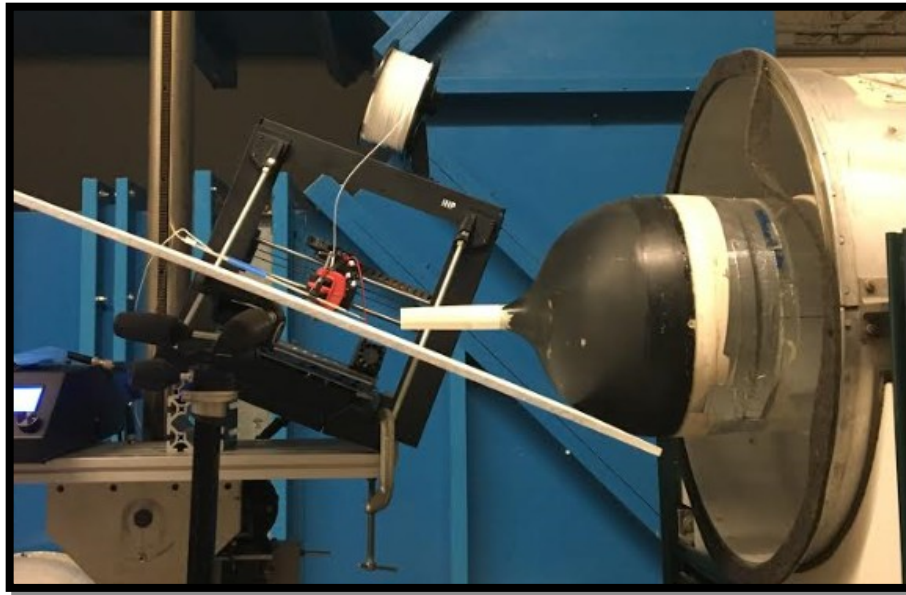
Nozzle 7



Nozzle 5

Nozzle Reference Name	Mechanical Swirl (Pitch)	Swirl Number
Nozzle 3	3.00"	0.31
Nozzle 5	5.00"	0.59
Nozzle 7	7.00"	0.71

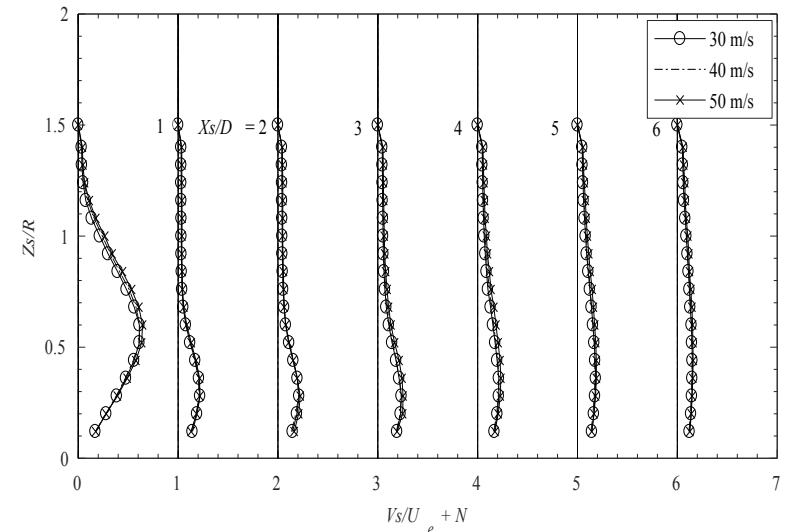
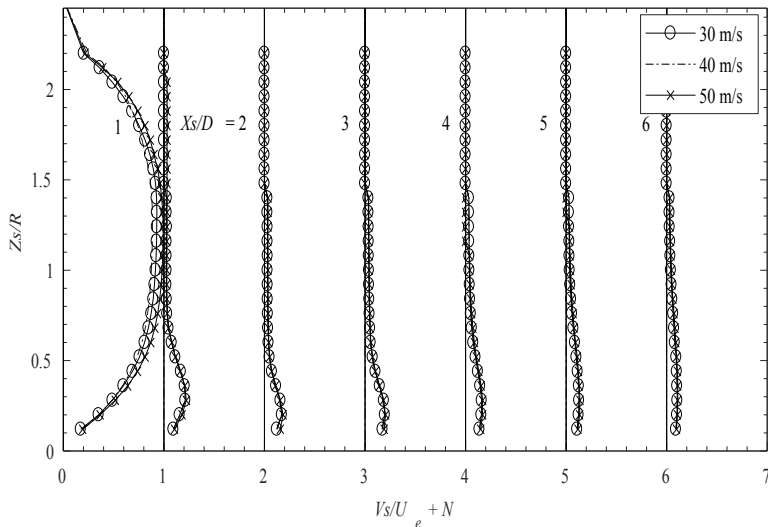
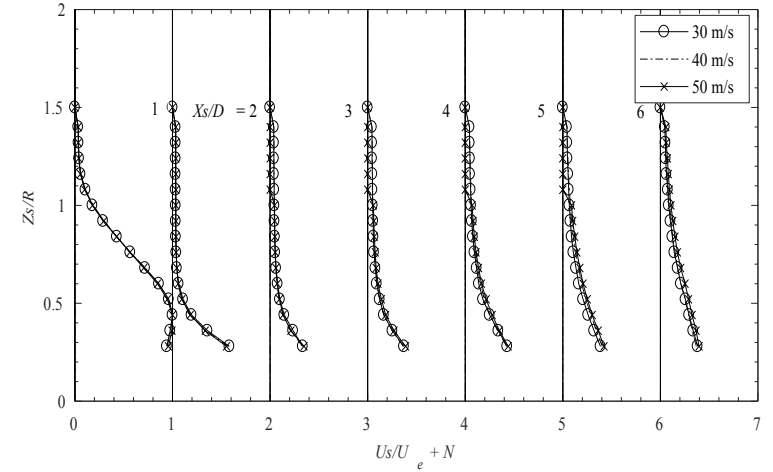
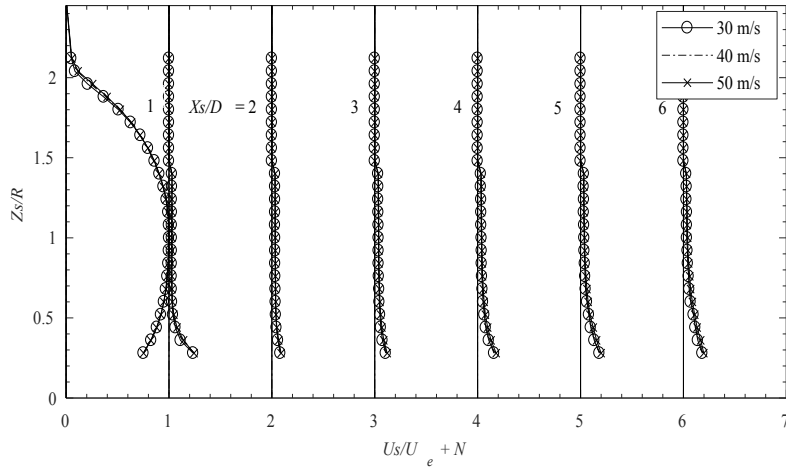
# Tripped Jet on Impingement Surface



Hot Wire Impingement Setup

- Distance from nozzle to impingement wall ( $d$ ) = 2.00"
- Impingement wall set to three angles of  $\theta$ : 22.5°, 45° & 67.5°
- Axial and azimuthal jet stream profile measured at 7 stations spaced two inch apart;
- Measured for three Reynolds numbers , 4.21 x 10<sup>4</sup>, 5.6 x 10<sup>4</sup>, 7.02 x 10<sup>4</sup>

# Tripped Nozzle Velocity Comparison

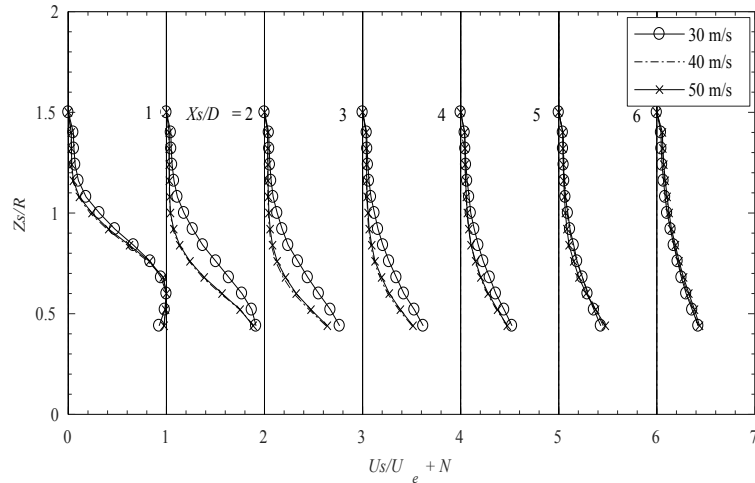


22.5° Axial and Azimuthal Profiles

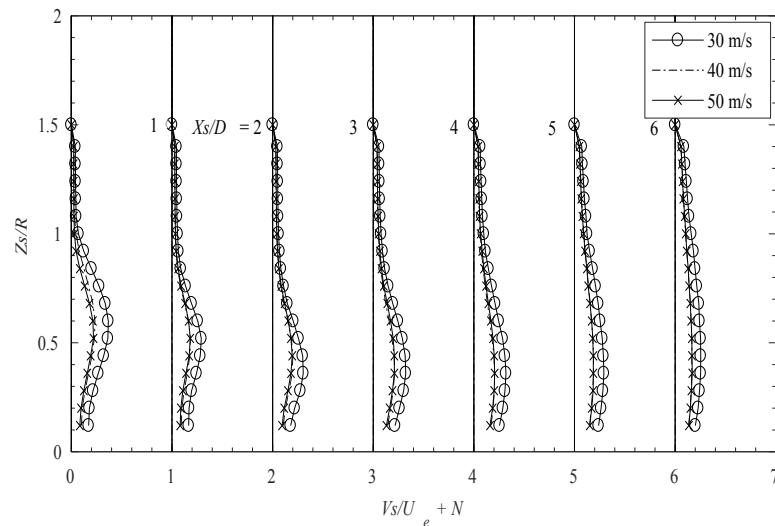
45° Axial and Azimuthal Profiles



# Tripped Nozzle Velocity Comparison

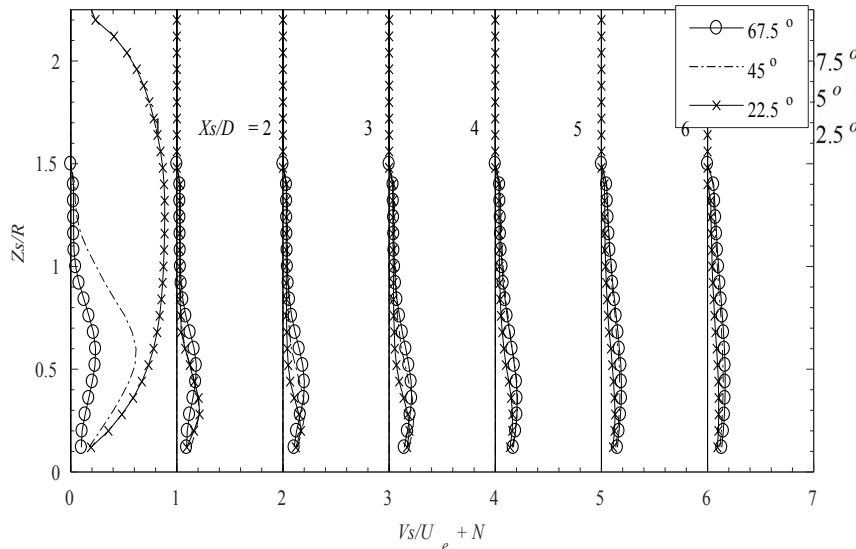
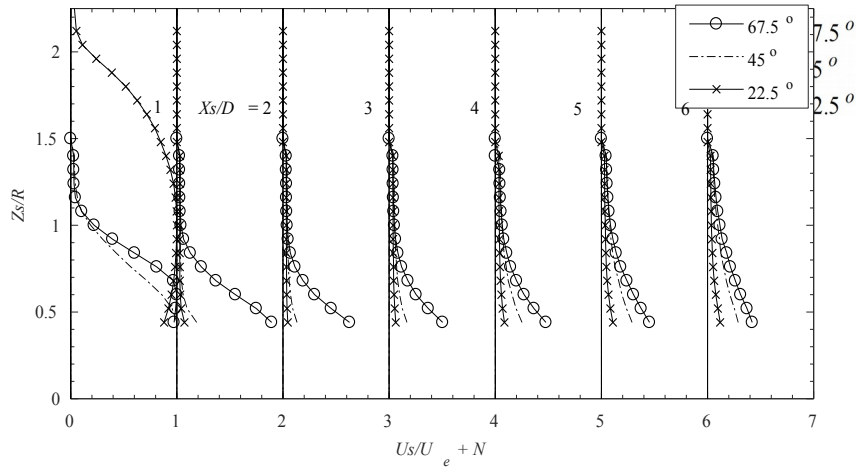


- Change in velocity does not influence the jet stream profile
- One outlier case of 50m/s at  $\theta = 67.5^\circ$
- Expected result because unimpeded flow shows similar characteristics



67.5° Axial and Azimuthal Profiles

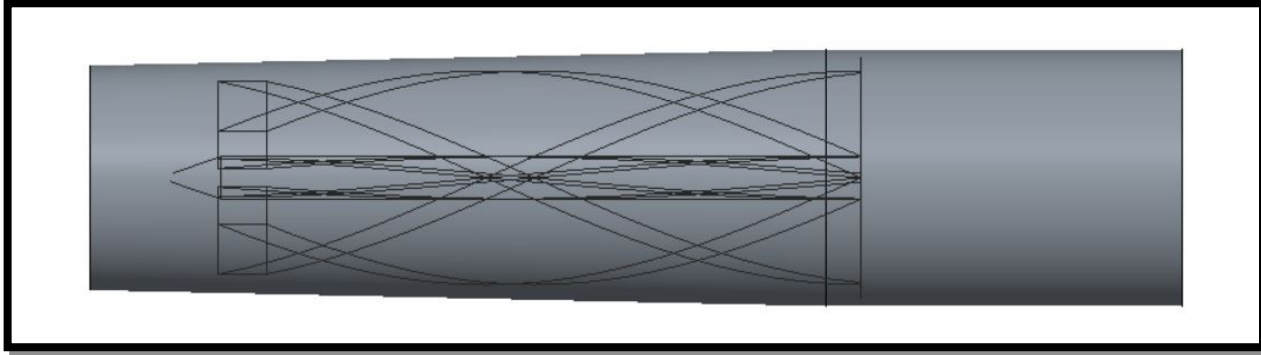
# Tripped Nozzle Angle Comparison



- Axial profiles differ greatly between setups
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  mildly effected by impingement wall
  - $\theta = 67.5^\circ$  nearly unaffected by impedance
- Besides station  $x = 0$  there is little change in the azimuthal profile
- Sufficient dissipation will be classified as a reduction in axial or azimuthal velocity by 50% before reaching station 2

40 m/s Axial and Azimuthal Profiles

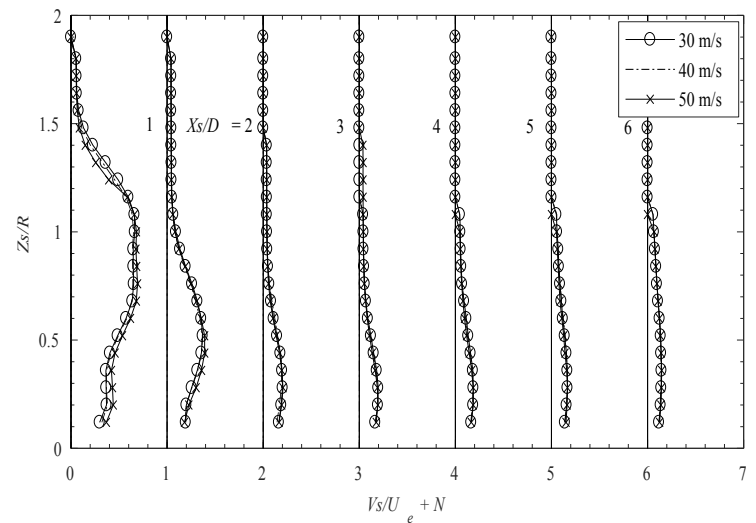
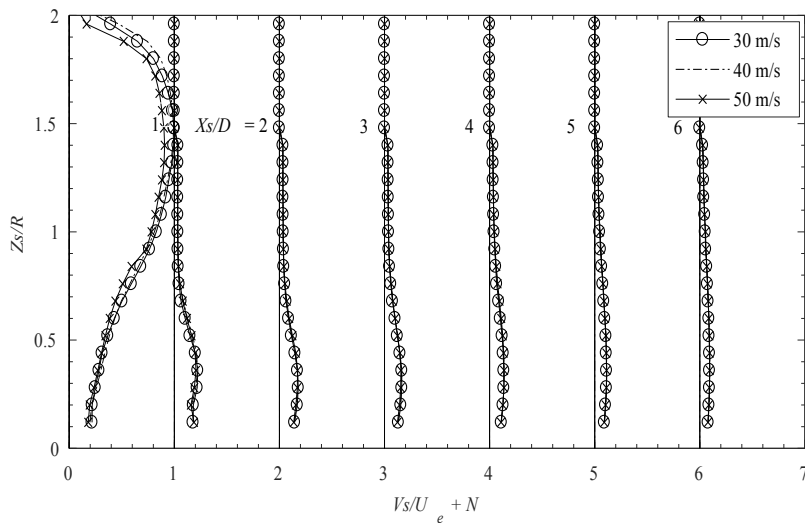
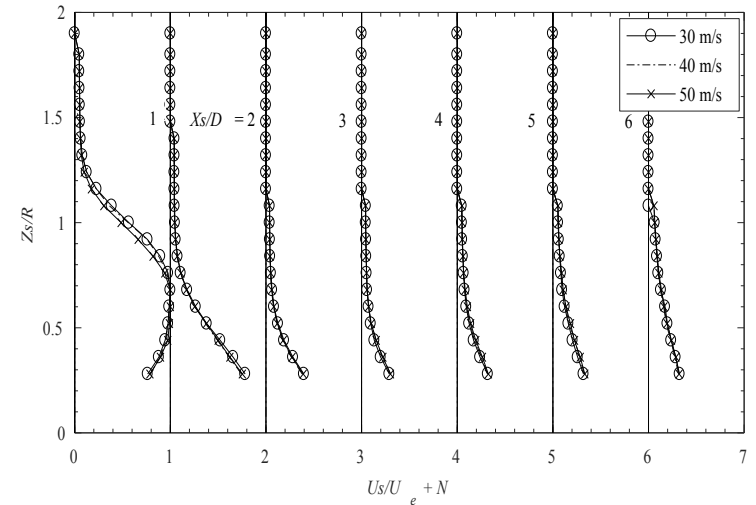
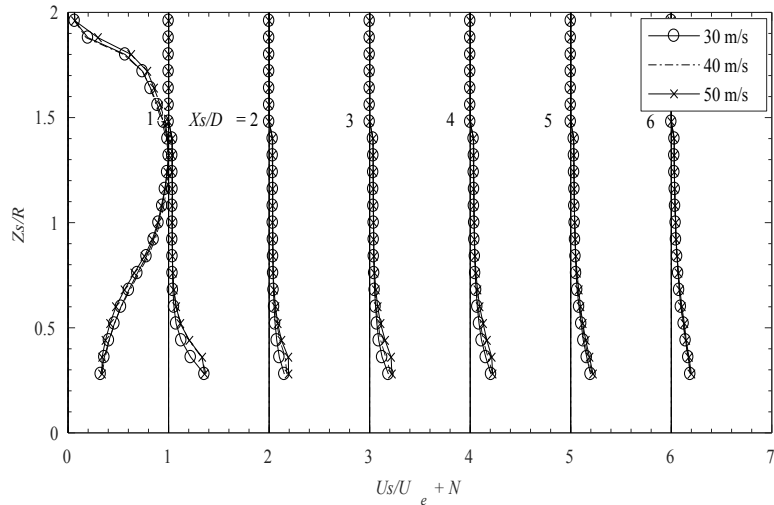
# Swirling Jet on Impingement Surface



Swirling Jet Stream Nozzle

- Three swirling nozzles; 3", 5" & 7" pitch
- Distance from nozzle to impingement wall ( $d$ ) = 2.00"
- Impingement wall set to three angles of  $\theta$ : 22.5°, 45° & 67.5°
- Axial and azimuthal jet stream profile measured at 7 stations spaced two inch apart;
- Measured for three Reynolds numbers ,  $4.21 \times 10^4$ ,  $5.6 \times 10^4$ ,  $7.02 \times 10^4$

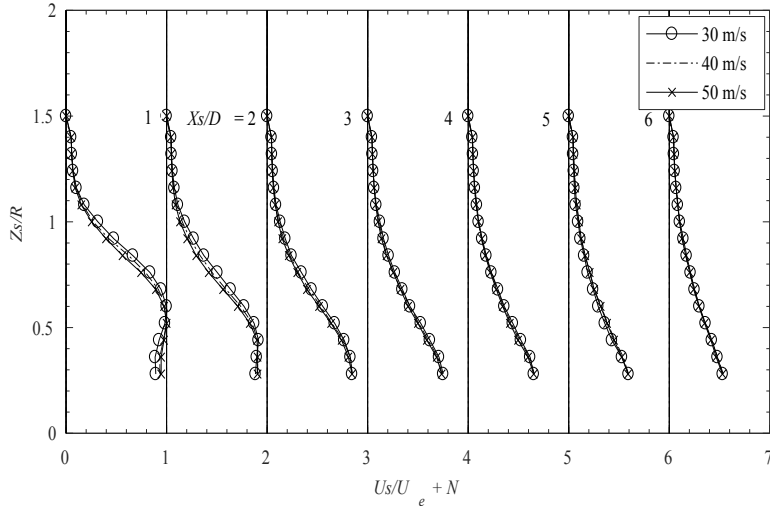
# Nozzle 3 Velocity Comparison



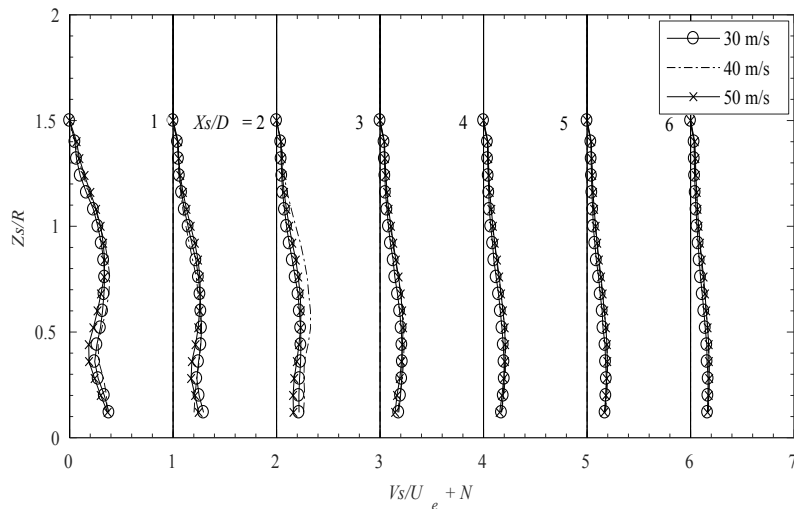
22.5° Axial and Azimuthal Profiles

45° Axial and Azimuthal Profiles

# Nozzle 3 Velocity Comparison

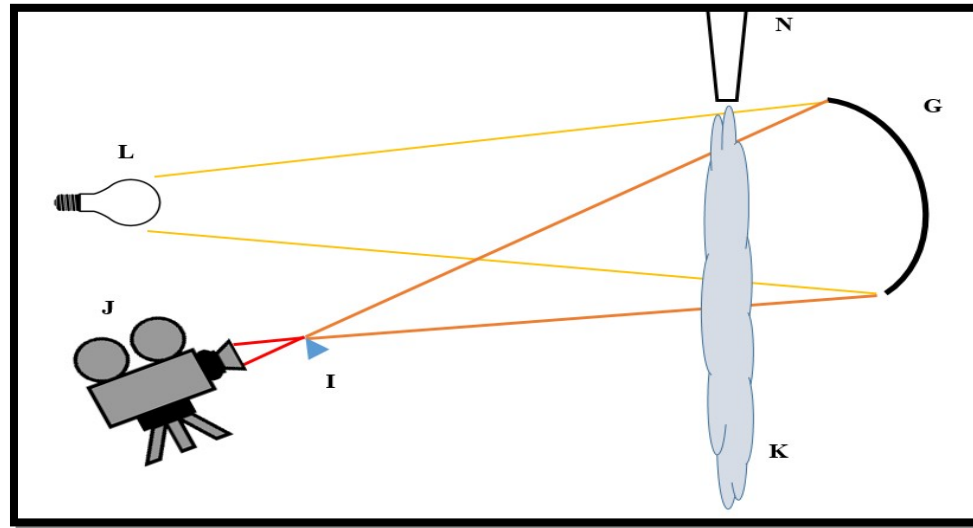


- Change in velocity does not influence the jet stream profile
- No significant outliers for Nozzle 3
- Nozzles 5 and 7 did have a few outliers like the tripped nozzle



67.5° Axial and Azimuthal Profiles

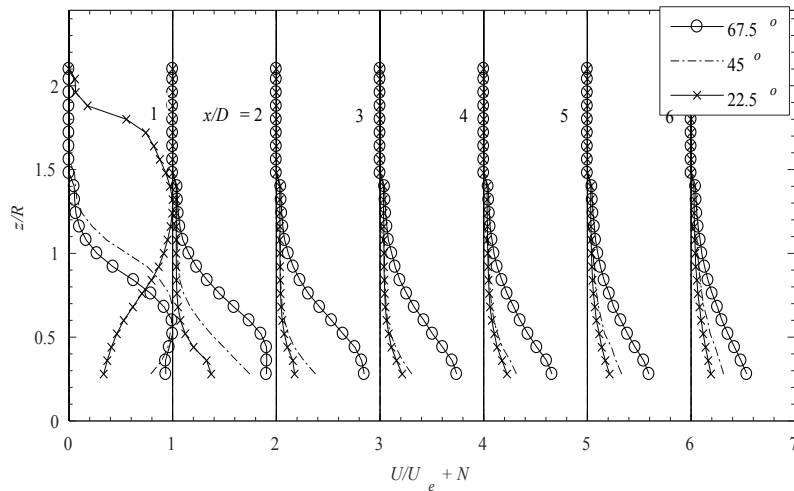
# Schlieren Images and Video



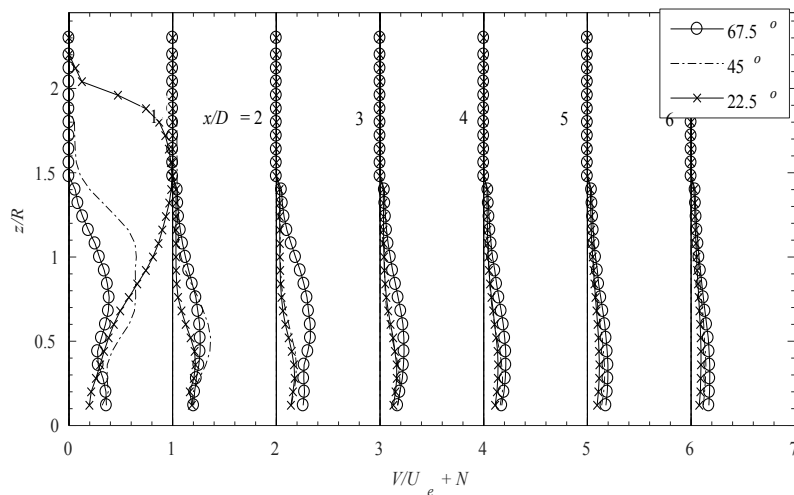
Swirling Jet Stream Nozzle

- Three swirling nozzles; 3", 5" & 7" pitch
- Distance from nozzle to impingement wall ( $d$ ) = 2.00"
- Impingement wall set to three angles of  $\theta$ : 22.5°, 45° & 67.5°
- Axial jet stream profile captured
- Measured for three Reynolds number  $5.6 \times 10^4$
- CO<sub>2</sub> seeding at 30psi, approximately a 50% increase in density

# Nozzle 3 Angle Comparison



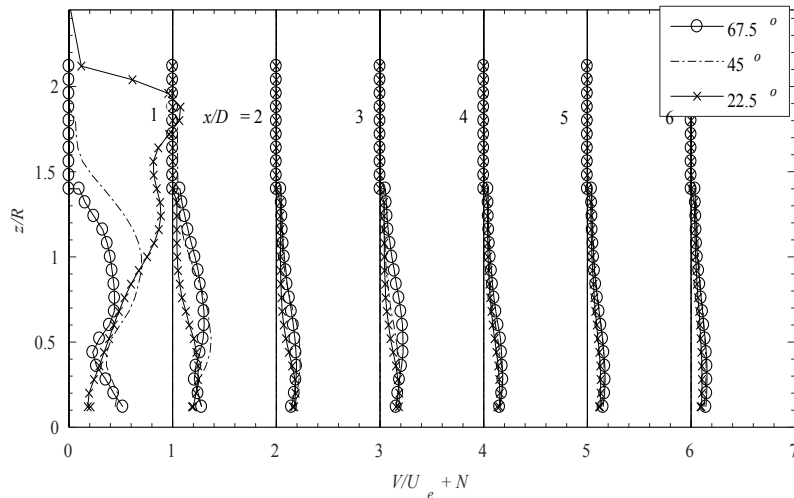
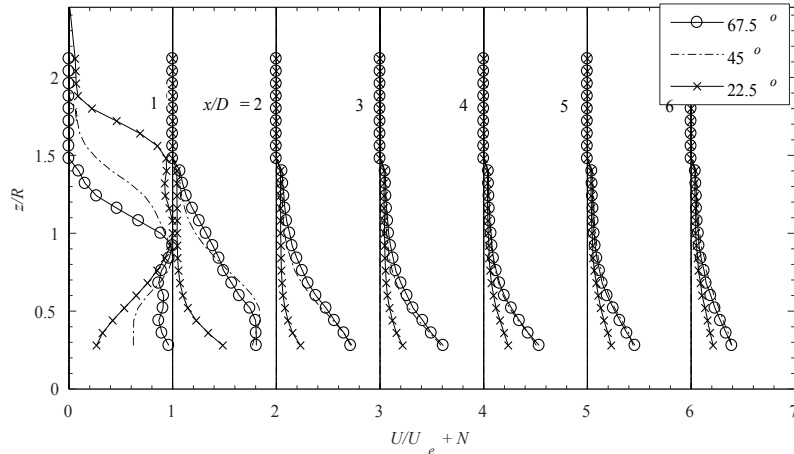
- Axial profiles differ greatly between setups
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  slows down but at half the rate of  $22.5^\circ$
  - $\theta = 67.5^\circ$  nearly unaffected by impedance



- Azimuthal profile differ greatly at  $x/D = 0$ 
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  slows down but at half the rate of  $22.5^\circ$
  - $\theta = 67.5^\circ$  effected but takes four times as long than  $22.5^\circ$

40 m/s Axial and Azimuthal Profiles

# Nozzle 5 Angle Comparison



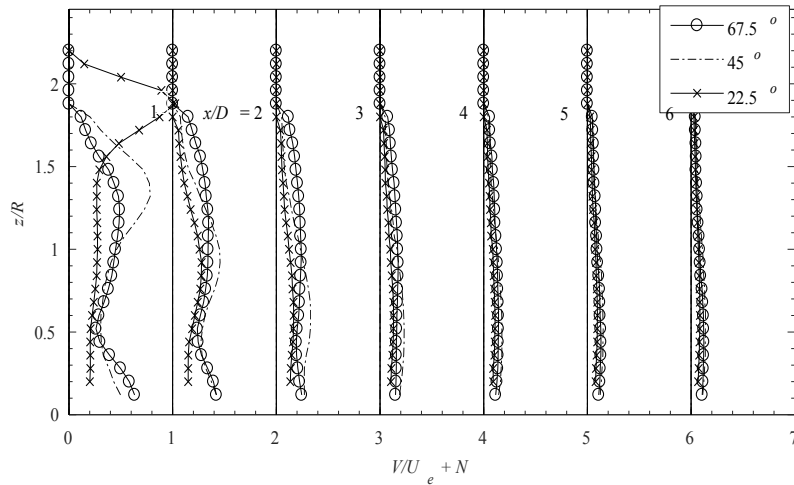
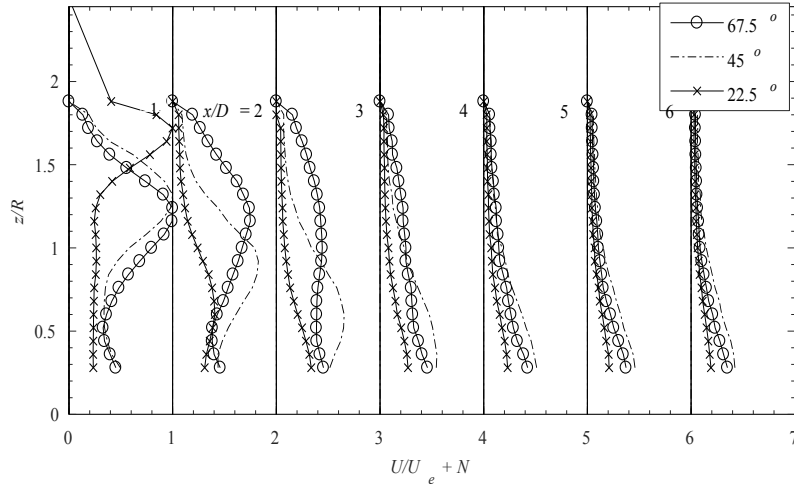
- Axial profiles differ greatly between setups
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  &  $67.5^\circ$  slows at about the same rate
  - Impingement surface is not as effective on the Nozzle 5 axial profile

- Azimuthal profiles differ greatly at  $x/D = 0$ 
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  slows down but at half the rate of  $22.5^\circ$
  - $\theta = 67.5^\circ$  is affected but takes four times as long as  $22.5^\circ$

40 m/s Axial and Azimuthal Profiles



# Nozzle 7 Angle Comparison

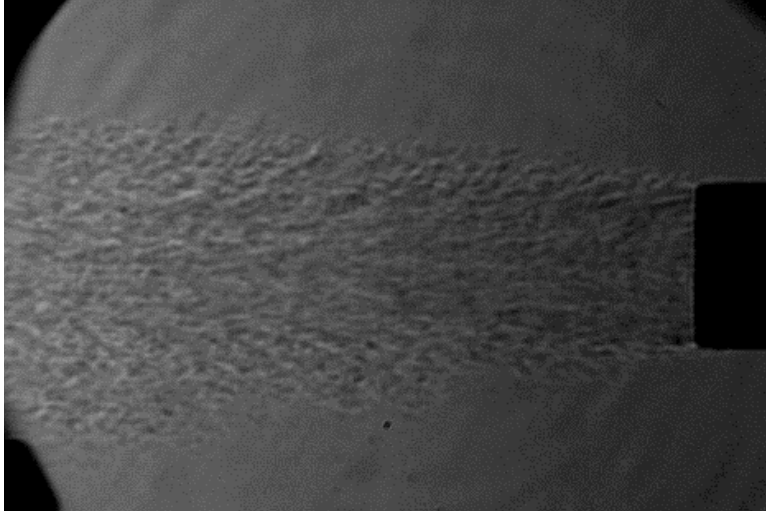


- Axial profiles differ greatly between setups
  - $\theta = 22.5^\circ$  dissipates quickest
  - $\theta = 45^\circ$  &  $67.5^\circ$  slows at about the same rate
  - Impingement surface least effective on the Nozzle 7 axial profile

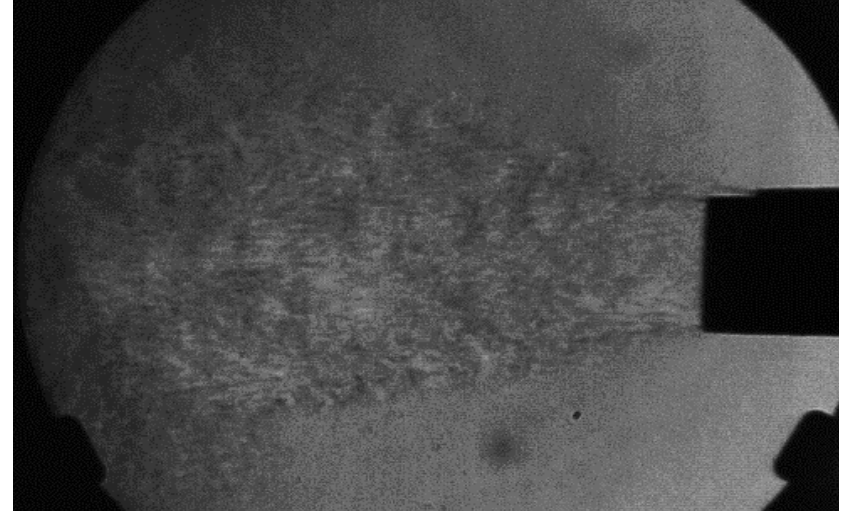
- Azimuthal profile differ greatly at  $x/D = 0$ 
  - $\theta = 22.5^\circ$  dissipates quickly
  - $\theta = 45^\circ$  slows down but at half the rate of  $22.5^\circ$
  - $\theta = 67.5^\circ$  effected but takes four times as long than  $22.5^\circ$

40 m/s Axial and Azimuthal Profiles

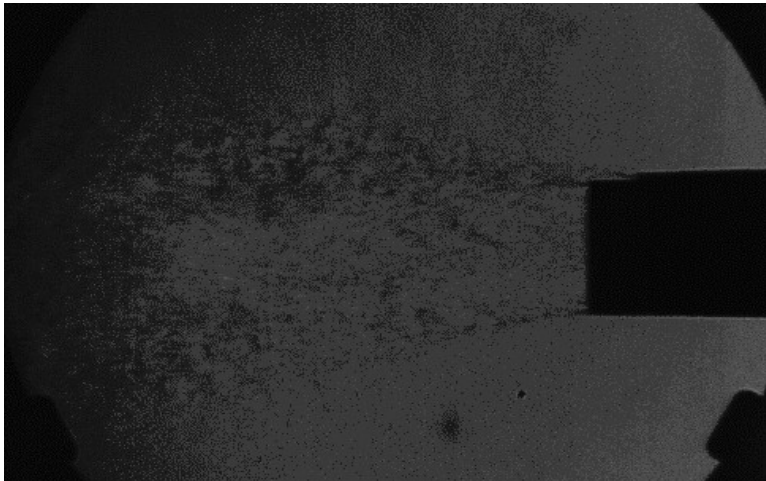
# Unimpeded Schlieren Images



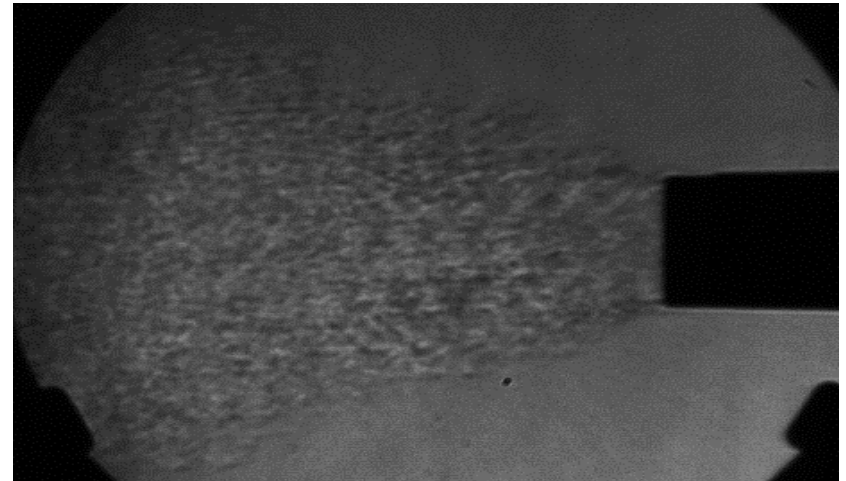
Tripped Nozzle



Nozzle 5



Nozzle 3



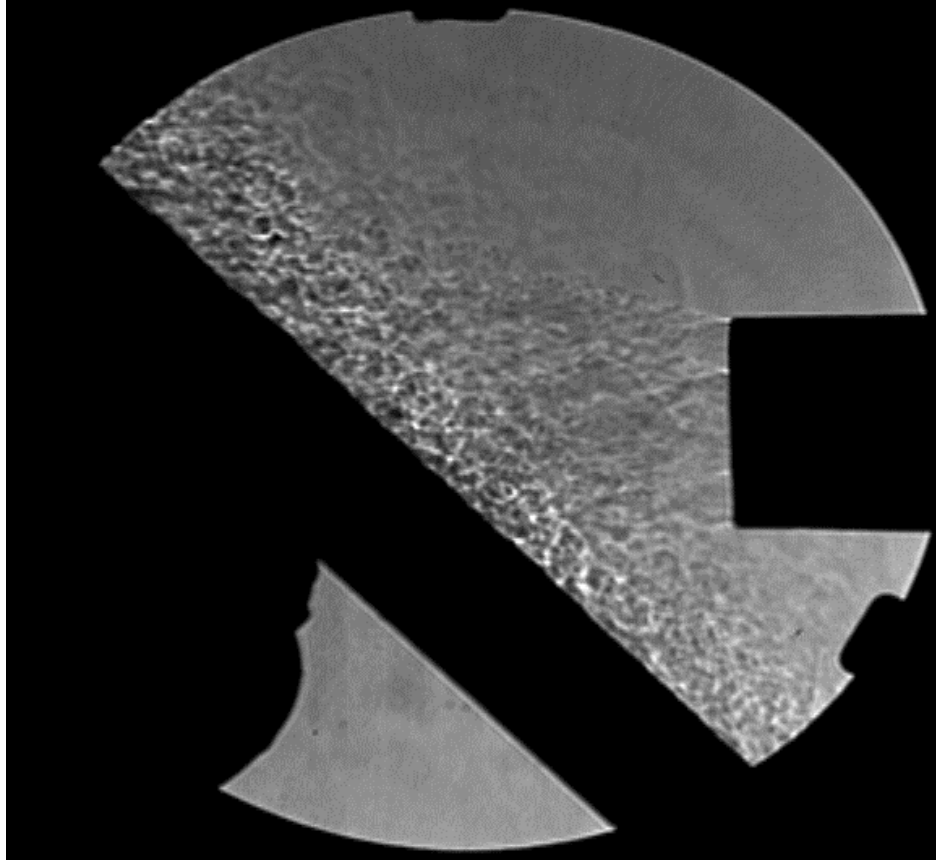
Nozzle 7

# Schlieren Video



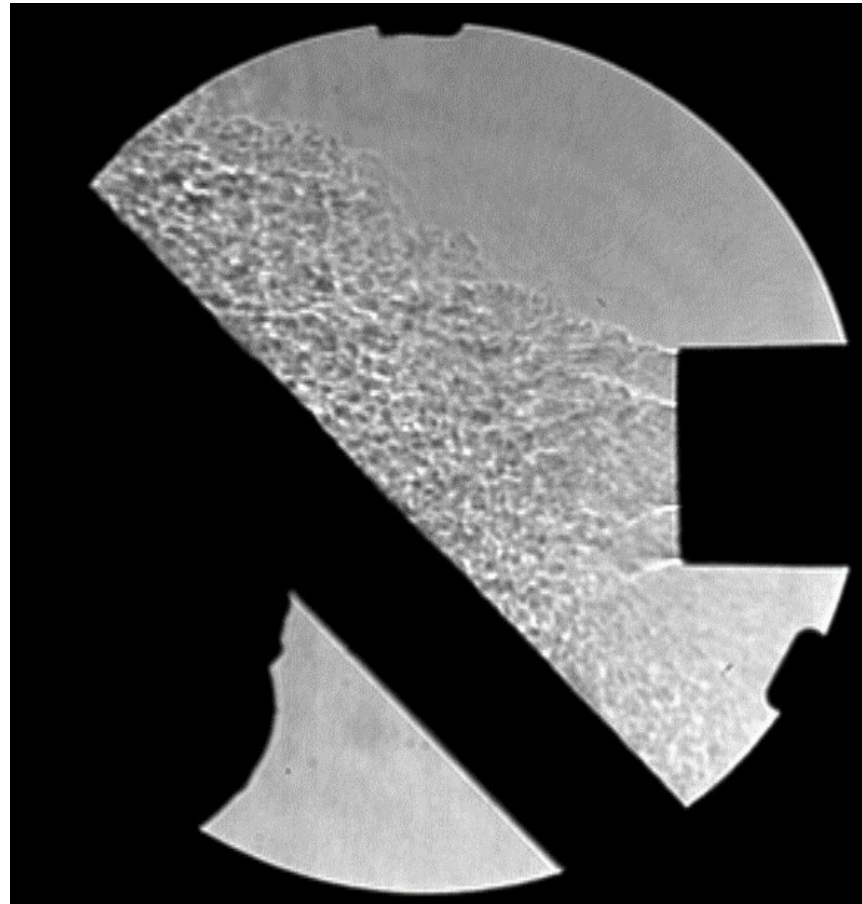
- Nozzle 3 with impingement wall at  $45^\circ$
- Vortex below centerline

# Schlieren Video



- Nozzle 5 with impingement wall at  $45^\circ$
- Small vortex below centerline

# Schlieren Video



- Nozzle 7 with impingement wall at  $45^\circ$
- Azimuthal vortex above centerline

# Summary of Results

- Angle required to dissipate jet stream flow increases as swirl number increases
- Required angle to effectively dissipate jet stream
  - $S = 0.31; \theta < 67.5^\circ$
  - $S = 0.59; \theta \leq 45^\circ$
  - $S = 0.71; \theta < 45^\circ$
- Azimuthal vortex formed when swirling jet stream was impinging on wall with  $\theta \geq 45^\circ$  and  $S = 0.71$
- Axial vortex formed below nozzle centerline when swirling jet stream was impinging on wall with  $\theta \geq 45^\circ$  and  $S \leq 0.59$

# Conclusions

- Reynolds number did significantly effect the vortex jet stream profile
- Dissipating vortex flow would require impingement walls to be set at high angles of attack to be effective
- Some interesting flow characteristics were observed for higher swirl number flows
  - Other forms of flow visualization could be used to better understand these behaviors such as oil field [2]

Questions?